| | Question Pa | per Code: 9970 | 2 | | | | | | | |
|--|-------------------------|-------------------------|--------------|--------------|-----------|--|--|--|--|--|
| B.E./B.Tech. DEGREE EXAMINATION, APRIL 2025 | | | | | | | | | | |
| Seventh Semester | | | | | | | | | | |
| Mechanical Engineering | | | | | | | | | | |
| 19UME702 – FINITE ELEMENT ANALYSIS | | | | | | | | | | |
| (Regulation 2019) | | | | | | | | | | |
| Duration: Three hours Maximum: | | | | | | | | | | |
| | Answer A | LL Questions | | | | | | | | |
| PART A - $(10 \times 1 = 10 \text{ Marks})$ | | | | | | | | | | |
| 1 is a Numeric mathematical physics. | cal method for solv | ing problems of Eng | gineering ar | nd | CO1- U | | | | | |
| (a) Finite Element Analy | /sis | (b) Finite Eleme | nt Method | | | | | | | |
| (c) Both (a) & (b) | | (d) none of the a | lbove | | | | | | | |
| 2. The art of subdividing a sknown as | structure into a con | venient number of s | maller elen | nent is | CO1- U | | | | | |
| (a) Non – Structural Pro | (b) Structural Problems | | | | | | | | | |
| (c) Discretization of stru | (d) None of the above | | | | | | | | | |
| 3 is a force acting at a particular point which causes displacement. | | | | | | | | | | |
| (a) Traction force | (b) Body force | (c) Point load | 1 (| (d) None of | the above | | | | | |
| 4. Assemblage of bars is ca | lled | | | | CO1- U | | | | | |
| (a) Truss | (b) Beams | (c) Spring | (d) None | of the above | e | | | | | |
| 5. Linear Strain Triangular | Element has | _ number of nodes. | | | CO1- U | | | | | |
| (a) 3 | (b)6 | (c)12 | | (d) 24 | | | | | | |
| 6. In plane strain analysis | | | | | CO1- U | | | | | |
| (a) $\rho_z = 0$ | (b) $\gamma_{xz} = 0$ | (c) $\gamma_{yz} = 0$ | | (d) All of t | the above | | | | | |
| 7 A motion which repeats i | itself after equal int | terval of time is calle | ed | | CO1- U | | | | | |

(b) Frequency (c) Counter flow

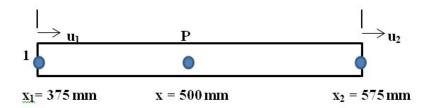
(d) Damping

(a) Cycle

Reg. No.:

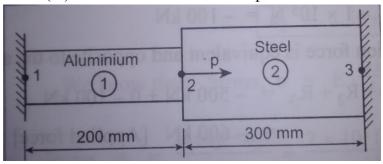
| 8 | 3 The causes of vibration is/are | | | | | | СО | 1- U | | | | |
|-----------------------------|--|--|---|-----------|----------------------------------|--------------|----------------------|------|--|--|--|--|
| | (a) W | inds | (b) Earthquakes | | (c) Elastic Nature | e (d) | (d) All of the above | | | | | |
| 9 | In nor | non-structural problems at each nodal point is obtained CO | | | | | | 1- U | | | | |
| | (a) Di | splacement | (b) Temperature | ; | (c) Stress | (d) | (d) Strain | | | | | |
| 10 | | is imag | inary line that con | nects a | series of points | | CO | 1- U | | | | |
| | (a) Path Line (b) Stream Line (c) Inviscid Flow (d) | | | | | | None of the a | bove | | | | |
| PART - B (5 x 2= 10Marks) | | | | | | | | | | | | |
| 11 | CO1- U | | | | | | | | | | | |
| 12 | 12 Explain Degrees of freedom. | | | | | | CO1- U | | | | | |
| 13 | 13 Write down the stress-strain relationship matrix for plane strain condition. | | | | | | | J | | | | |
| 14 | 14 State difference between Direct and Iterative methods for solving system of equations. | | | | | | | | | | | |
| 15 | 5 Write down the expression for stiffness matrix in 2D fluid mechanics. | | | | | | CO1- U | J | | | | |
| PART – C (5 x 16= 80 Marks) | | | | | | | | | | | | |
| 16 | phenomenon. $\frac{d^2y}{dx^2} + 50 = 0 \qquad , 0 \le x \le 10$ and the trial function is $y = a_{IX}(10-x)$ with boundary conditions as $y(0) = 0 \text{ and } y(10) = 0. \text{ Find the value of the parameter } a_I \text{ by the following methods.}$ (i) Point Collocation Method (ii) Sub-domain Collocation Method (iii) Least Squares Method (iv) Galerkin's Method Or | | | | | | | | | | | |
| | (b) | concentrated | B of span L simp load W at the certing Rayleigh Rit | ntre C. A | Analyze the deflec | tion at mid- | CO4- Alla | (10) | | | | |
| 17 | (a) | $750 \text{mm}^2 \text{ and } u_2 = 0.625 \text{mm}^2$ | | is is 2 X | 10^5 N/mm^2 . If u_1 | | CO2- App | (16) | | | | |

(v) Element Stiffness Matrix [K]



Or

- (b) An axial load of $4x10^5$ N is applied at 30° C to the rod as shown in CO2-App (16) the figure. The temperature is then raised to 60° C. Calculate the following:
 - (i) Assemble the K and F matrices
 - (ii) Nodal Displacements
 - (iii) Stresses in each material
 - (iv) Reactions at each nodal point



18 (a) For the plane strain element as shown in figure the nodal displacements are

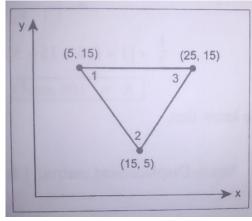
CO2- App (16)

$$u_1 = 0.005 \text{ mm}; \quad v_1 = 0.002 \text{ mm}$$

$$u_2 = 0 \text{ mm}; v_2 = 0 \text{ mm}$$

$$u_3 = 0.005 \text{ mm}; \quad v_3 = 0 \text{ mm}$$

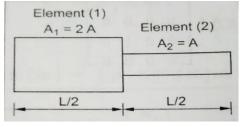
Determine the element stresses σ_x , σ_y , τ_{xy} , σ_1 and σ_2 and the principal angel Θ_p . Let E=70Gpa, v=0.3 and use unit thickness. All coordinates are in mm



- (b) (i) Evaluate the integral $\int_{-1}^{1} \frac{\cos x}{1-x^2} dx$ by applying 3 point Gaussian Quadrature. CO2- App (8+8)
 - (ii) Evaluate the integral $\int_{-1}^{1} (2 + x + x^2)$ dx and compare with exact solution.

Or

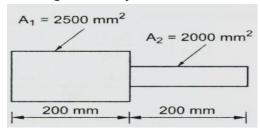
19 (a) Compare the natural frequencies of longitudinal vibration of the CO4- Ana (16) unconstrained stepped bar as shown in the figure.



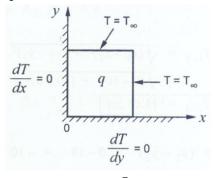
Or

(b) Compare the eigen values and frequencies for the stepped bar as CO4- Ana (16) shown in the figure.

Take, Young's modulus $E = 2 \times 10^5 \text{ N/mm}^2$, Unit weight Density = $0.8 \times 10^{-4} \text{ N/mm}^3$



20 (a) Evaluate the temperature distribution in a square region with CO6-Eva (16) uniform energy generation as shown in figure. Assume that there is no temperature variation in the z-direction. Take k=30W/cm°C, l=10cm, T=50°C, q=100W/cm³.



Or

(b) A steel rod of diameter d= 2 cm, Length L=5 cm and thermal CO6-Eva (16) conductivity k = 50W/m°C is exposed at one end to a constant temperature of 320°C. The other end is in ambient air of temperature 20°C with a convection coefficient of h = 100 W/m²°C. Evaluate the temperature at the midpoint of the rod.