

L1B
10/12/13 FN

Reg. No. :

Question Paper Code : 82322

M.E. DEGREE EXAMINATION, NOVEMBER/DECEMBER 2013.

Elective

Structural Engineering

ST 9258/UST 9158/10211 SEE 51 — MECHANICS OF COMPOSITE MATERIALS

(Regulation 2009/2010)

Time : Three hours

Maximum : 100 marks

Answer ALL questions.

PART A — (10 × 2 = 20 marks)

1. How are the composite materials classified?
2. What are the advantages of ceramic matrix composites?
3. State: Hooke's law.
4. Are V_{12} and V_{21} independent of each other for a unidirectional orthotropic lamina?
5. Define the major Poisson's ratio.
6. Distinguish: Specially orthotropic plies and generally orthotropic plies.
7. Is a nonzero [B] matrix attributed to the orthotropy of layers?
8. Determine the Tsai-Hill failure criteria for pure shear at various angles θ to the principal Material directions.
9. What are the applications of composites in various fields?
10. State whether natural fiber composites are environmentally superior to glass fiber reinforced composites?

PART B — (5 × 16 = 80 marks)

11. (a) (i) Give a brief historical review of composites. (8)
- (ii) Compare the flexibility of a 0.3 mm diameter steel wire to 0.6 mm diameter Aluminium wire. The young's modulus of steel is 200 GPa and that of aluminium is 70 GPa. (8)

Or

- (b) (i) Describe any one manufacturing method of polymer matrix composites. (8)
- (ii) A lamina consists of 100 fibers of $10\mu\text{m}$ diameter. The fibers are 10 mm long. Find the interfacial area. What is the increase in the interfacial area if the diameter of the fiber is reduced to $5\mu\text{m}$ and the total volume of fibers is kept constant? (8)
12. (a) A uniaxial load is applied to a 10° ply. The linear stress-strain curve along the line of load is related as $\sigma_x = 123 \varepsilon_x$, where the stress is measured in GPa and strain in m/m. Given $E_1 = 180$ GPa, $E_2 = 10$ GPa and $\nu_{12} = 0.25$. Find the value
- (i) shear modulus, G_{12} and (8)
- (ii) modulus E_x for a 60° ply. (8)

Or

- (b) (i) Reduce the monoclinic stress-strain relationships to those of an orthotropic material. (8)
- (ii) A lamina is loaded at angles $\theta = 30^\circ$ and 45° with the fiber direction, and the corresponding moduli obtained are E_{x30} and E_{x60} . Determine a relationship between E_1, E_2, E_{x30} and E_{x45} . Determine an expression for E_2 in terms of E_{x30} and E_{x45} for $E_1 \gg E_2$. (8)
13. (a) A cross-ply laminate $[0/90]_s$ made from high strength carbon/ epoxy unidirectional plies and subjected to a tensile membrane longitudinal force of $N_x = 100$ N/mm. Each ply is 0.125 mm thick and have identical properties as given below in Fig. 1. Determine the longitudinal stress and strain. (16)

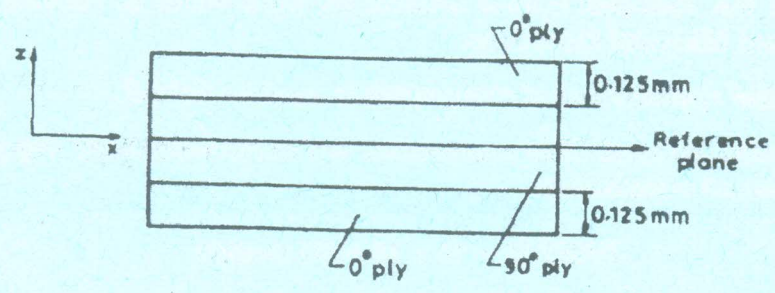


Fig. 1

Or

- (b) (i) Derive the expressions for the stiffness matrices [A],[B] and [D] for an isotropic material in terms of its thickness, t , Young's modulus, E , and Poisson's ratio, ν . (10)
- (ii) Show that for a symmetric laminate, the coupling stiffness matrix is equal to zero. (6)
14. (a) (i) An E-glass/epoxy composite has the following properties of its constituents
- $\nu_f = 0.65$ $E_{1f} = 70$ GPa $E_m = 3.5$ GPa
- $F_{1ft} = 3500$ Mpa and $F_{mt} = 100$ MPa
- Determine the longitudinal modulus of the composite E_1 and longitudinal tensile strength F_{1T} of the composite. (10)
- (ii) Find the Tsai-Hill failure criteria for pure shear at various angles θ to the principal material directions. (6)
- Or
- (b) (i) The loading axis is inclined at an angle θ to the fibre axis. Tension along they axis is the only load applied. Using maximum strain criteria, Determine F_{mt} and θ so that prediction of inplane shear and tensile failure load coincide. (8)
- (ii) A lamina is loaded as shown Fig 2. Using Tsai-Wu failure criterion, determine F_0 . (8)

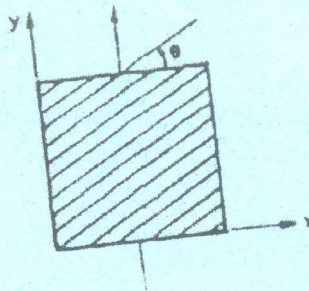


Fig 2

15. (a) (i) Explain the geometry and stress variation of adhesive joint with a neat diagram. (12)
- (ii) Write the advantages and disadvantages of Mechanical joints. (4)
- Or
- (b) (i) Describe about the mode of failure of mechanical joints. (8)
- (ii) What are the factors affecting the laminate selection for fibre reinforced polymer? (8)